Pure & Appl. Chem., Vol.52, pp.1865-1882.
Pergamon Press Ltd. 1980. Printed in Great Britain.

FROM POLYMERS TO POLYMERIC CARBON - A WAY TO SYNTHESIZE A LARGE VARIETY OF NEW MATERIALS

Erich Fitzer

Institut für Chemische Technik, University Karlsruhe, F.R. Germany

Abstract - Polymeric carbon became known during the last years in form of monolytic bulk material as glassy carbon and carbon fibres as well as in combination of both as carbon/carbon composites. The paper involves a general review of the nature of these polymeric carbons as a single element polymer with two-dimensional polymer units. The chemical viewpoints for the preparation of these polymeric carbons and their application in combination with organic polymers are treated in detail.

The present and future application possibilities are discussed basing on the material capacity, technological difficulties and economic viewpoints.

INTRODUCTION

Polymeric carbon is made from synthetic polymers. World polymer production amounts more than 40 Mio jato. The polymer production including fibres and elastomeres comprises a fifth of the total volume of the production of chemical industry throughout the world. All synthetic polymers are made of crude oil as raw material. Therefore, one has to examine whether this basic raw material for today's polymers and for special new products made from it, will be available in the future which will be characterized by a growing energy crisis. Or, have we to go back to the metals or ceramic materials instead of to proceed from organic polymers to the low weight anorganic material "Pure Carbon"?

ALUMINIUM		15.2
COPPER	10.7	
STEEL POLYAMIDES POLYFORMALDEHYDES STYROLCOPOLYMERE	78 4.2	
POLYSTYROLE POLYETHYLENE (HIGH DENSITY) POLYVINYLCHLORIDE	20 1.8 1.7	
POLYPROPYLENE POLYETHYLENE (LOW DENSITY)	16 1.5	

Fig. 1. Oil equivalents for the production of various materials, needed for energy and raw materials.

Referring to the newest study of chemical industry, Fig. 1 (Ref.1), the need for energy plus raw material as shown in equivalent units of crude oil is much lower for polymers than for the most common metals which last ones additionally need other raw materials than oil.

Such comparisons become impressive if we consider the final products for example: from 1 kg crude oil 300 m polyethylene pressure water tubes can be produced, but only 50 m of same sized tubes made of steel with protective coating. A comparison of waste water tubes shows more than 12 times superiority of PVC to cast iron. Such considerations became even more impressive if we compare mobile products for which additional energy is saved during application by light weight materials (compare Fig.22).

We can conclude that we have not to go back to the metal-age, if we save our fossile fuels as raw material for producing most valuable materials instead of simple burning off.

THE BASIC PROPERTIES OF PURE CARBON AS MATERIAL

The outstanding properties of pure carbon materials if compared with organic polymers are first of all the unlimited thermostability, the high corrosion resistance, and their absolute biocompatibility.

Disadvantages are the poor processibility, and as indirect consequence their poor mechanical properties and their brittle fracture behaviour. Conventional carbon and graphite products must be manufactured like ceramic materials. The heterogeneous structure consists of granular particles and binder bridges with an unavoidable volume fraction of pores in the order of at least 20%. Such a heterogeneous structure does not utilize by far the superiour strength of the chemical bonds between the carbon atoms. For better understanding of the bond strength, we must remember the electron configuration of the carbon atoms in compounds namely

- 1) the sp³ hybridization, resulting in four equivalent uncompletely filled hybride orbitals, and
- 2) the formation of only three equivalent sp 2 hybrides.

In the last case there remains one original p-orbital which is occupied by the remaining valence electrons. These p-electrons are forming the π -bonds which are weaker than the σ -bonds but contribute to the strength of the three covalent bonds. The total strength of one of these three bonds is 430 kJ/mol, whereas each of the four equivalent bonds in the tetrahydral structure has a strength of 350 kJ/mol only.

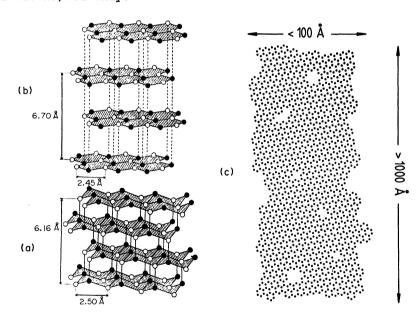


Fig. 2. Structures of elemental carbon :
 a) diamond, b) graphite, c) polymeric carbon

Both types of chemical bonds are also found between the atoms of elemental carbon. The $\rm sp^3$ hybridization in the diamond modification (Fig.2a) and the $\rm sp^2$ hybridization in the graphite structure (Fig.2b), but also in the imperfect graphitic carbons consisting of layers of different sizes with more or less distortion (Fig. 2c). Typical for these two alternative hybridizations are the different bond angles of 120°C for $\rm sp^2$ and 109° for $\rm sp^3$, as well as the different bond lengths and lattice distance. All bonds are strictly orientated. The $\rm p_z$ -orbitals of the $\rm sp^2$ hybridization are oriented perpendicular to the layer planes. In the perfect graphite lattice however, the wave function of these p-electrons is more extended in space and not localized between two atoms only. This causes mobility of the valence electrons like that in the most metallically bonded elements.

A very weak bond in c-direction of the graphite structure is the consequence. In disordered graphitic carbons, some cross-linking between the layers and limited delocalization of the p-electron is observed.

THE NATURE OF POLYMERIC CARBON

It was known since many years that also structurally disordered carbons, formerly defined as "amorphous carbon", consists already of small units of graphitic layers, however, without any crystalline order in the third dimension. What was not known,however, is the new finding that there exist disordered carbons which have structural polyaromatic units of strip- or ribbonlike shape and which are much larger than indicated by wide angle X-ray diffraction measurements. As shown in Fig. 2c, such planar macromolecules are extended up to 100 Å in width, corresponding to a sequence of at about 80 carbon atoms, but are practically infinitive in length. Only the technique of high resolution electron transmission microscopy has revealed the structural arrangement of the ribbons and plates. This molecular structure reminds of the organic polymers with the only difference that not chain- or ladder-molecules but planar molecules are the structural units. This form of carbon should therefore be named "Polymeric Carbon" (Ref. 2).

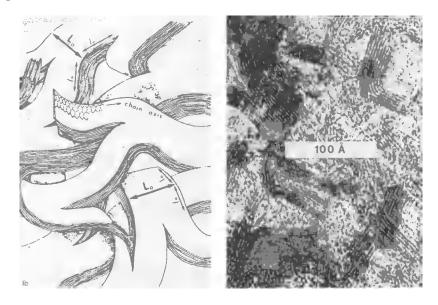


Fig. 3. Structural model of polymeric carbon (left) based on high resolution TEM (right) (Ref. 4).

In polymeric carbons also some of such polyaromatic layers are arranged in parallel positions but without forming a defined lattice in respect to the third dimension. The polyaromatics are imperfect themselves. There exists a strong linkage between the layers, and also some cross linkage between the plates or ribbons formed by these parallel arranged polyaromatic layers. Therefore polymeric carbon can not be transformed into a graphite structure by heat treatment above 2600 °C. It is a so-called "non-graphitizing carbon", that means the p-electrons are not delocalized and even sp³ - hybridization can be assumed in the cross linkages between the polyaromatic molecules. But never, tetrahydral structure units have been confirmed by X-ray diffraction so far(Ref.3).

The structural model of isotropic polymeric carbon as shown in Fig.3 (Ref.4) has been developed by TEM studies, but is also supported by improved X-ray - diffraction and adsorption measurements (Ref. 5).

As far as the realization of bond strength in a bulk material is concerned, we can expect that the two-dimensional strength within the sheets is distributed statistically within an isotropic bulk material. If we exclude plastic deformability, high strength but brittle fracture behaviour should be observed. Such an isotropic material became known 20 years ago as "glassy or vitreous carbon". It is made from polyfurfuryl alcohol or phenolics by careful coking. Due to

the analogy with the morphology of the precursor polymer, polymeric carbon is a monolytic carbon without any grain boundaries. As shown in Fig. 4 the strength of such isotropic monolytic carbon materials exceeds with 200 MN/m considerably that of conventional polygranular graphites. Only newer types of ultrafine grained graphite exhibit comparable strength properties.

The striking effect of strength increase was achieved by preferred orientation of the planar polymeric molecules. As shown in Fig. 4 tensile strength up to $4000~\text{MN/m}^2$ can be obtained in direction parallel with the layers. This second type of monolytic polymeric carbon is realized in carbon fibres.

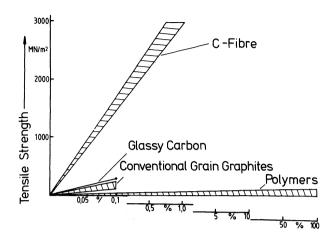
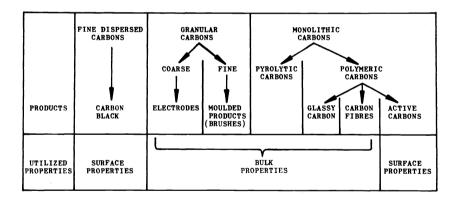


Fig. 4. Stress/strain-behaviour of various carbon materials

For better understanding table 1 shows where the polymeric carbons are located within the variety of carbon and graphite materials. We can distinguish between fine dispersed carbons (the carbon blacks), the granular carbons (the electrodes made by a ceramic like process), and the monolytic carbons.

TABLE 1. Variety of carbon- and graphite materials



Besides of the above mentioned polymeric carbons, monolytic carbon can also be prepared by chemical vapour deposition (pyrolytic graphite). It is discussed later that carbon fibres can be imbedded in a carbon isotropic matrix like glassy carbon. Such heterogeneous structure is known as carbon fibre reinforced carbon (CFRC), or simply carbon/carbon composites (CCC).

Table 2 (Ref.6) gives a survey on the properties which can be achieved with the various types of structural carbons and especially on the yield of theoretical mechanical properties in solid carbon. The yield of theoretical YOUNG's modulus is derived from calculations of the bond strength. The anisotropic type of monolytic carbons can achieve up to 50 % yield of the theoretical YOUNG's modulus. The estimation of a theoretical strength is based on the assumption that fracture does not start from cracks but occur simultaneously between all atoms of two neighbouring layers. For graphite the maximum theoretical strength is at about 1:10 of the YOUNG's modulus in a first approximation.

	MONOLITHIC CARBON			HETEROGENOUS CARBON		
	GLASS LIKE	PYROLITIC	C-FIBRE	C/C - COMPOSITE	CARBON ultra fine coarse grained grained tyne needle type	
YOUNGs MODULUS	¹ 30	// 30	# ³⁰⁰⁻ 500	400	10 -12	
TENS/FLEX STRENGTH MN/m ²	// 120	# 15 0	2000- 4000	1000	10 - 80	
COMRESSIVE STRENGTH MN/m ⁻²	# 350	1 450			50-200	
YIELD OF THEORETICAL YOUNGS MODULUS	1 (3%	<i>II</i> <5 %	30 ₋ 50	50%	<0,2%	
YIELD OF THEORETICAL STRENGTH	(25	(5	~ 10	5	<0,1	

(0.002%

TABLE 2. Yield of theoretical mechanical properties in the various all-carbon materials

THE PREPARATION OF POLYMERIC CARBON BY CONTROLLED THERMAL DEGRADATION OF SYNTHETIC POLYMERS

(0,1%)

If one takes a natural polymer like wood or jute fibres or coconut shells, the thermal degradation will result in a porous carbon material, in which the cellular structure of the precursor is completely preserved.

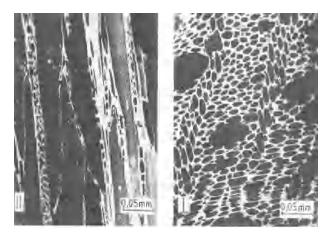


Fig. 5. Optical micrographs of wood-char

Fig. 5 shows the micrographs of a wood char in cross section parallel with and perpendicular to the wood structure. One recognizes by this very porous structure that the structural reorganization of the carbon atoms remaining after carbonization can be realized only in limited areas which have existed before as macro-homogeneous polymer-structures. In the cellulose structure, e.g., these limited areas are fibrillous (Ref. 7). In the case of wood these are the walls of the cells.

The mass loss by volatile byproducts causes a considerable shrinkage. This is easy to understand from the formula of cellulose, and a comparison with the structural arrangement within polyaromatic planes, present in the carbon residue (Fig. 6). The low carbon yield of less than 30 % and the porous structure cause extreme fragility. The mechanical properties can be improved by use of porefree semi-synthetic polymers as precursor, f.i. rayon. Rayon was the raw material for the first carbon fibres in the sixties (Ref. 8), and is still used as raw material for commercially available carbon felt.

Fig. 6. Structural rearrangement of the C-atoms during pyrolysis of cellulose
Fig. 7. Formation of polymeric carbon from polyfurfuryl alcohol (Ref. 10,11)

The mechanical properties of carbon fibres from rayon are some orders of magnitude higher than the very first extremely brittle carbon fibres produced by EDISON, 100 years ago, to be used in electrical bulbs, although at that time already some hundred thousands of gold dollars have been spent for finding a better suitable natural precursor material.

A much more perfect formation of the planar, polyaromatic macromolecules which form the polymeric carbon is achieved by use of thermosetting resins as precursor. In this way a carbon yield up to 50 % and more can be achieved. A linear shrinkage up to 25 % is reported for polyfurfurylalchohol and for phenolics as raw material (Ref. 9). The pyrolysis chemistry of polyfurfurylalcohol and the formation of the carbon sheets out of it are shown in Fig. 7. Most of non-carbon atoms are splitted off below 450 °C as volatile byproducts (Ref.10). Above this temperature a high amount of shrinkage is observed. Very carefull and highly sophisticated X-ray diffraction studies (Ref.11) have shown how the planar polyaromatic sheets are formed by polymerization reactions from the linear carbon chains as intermediates. It is assumed that the growth of these carbon chains controlls the length of the carbon ribbons.

Fig. 8. Thermal degredation of p-polyphenylene (Ref. 12)

From chemical viewpoint p-polyphenylene would be an ideal precursor for polymeric carbon because of its high carbon yield of more than 80 % during thermal degradation.

The carbonization chemistry (Fig.8) is so interesting because one can detect the defect structure of the sheets between 600 and 700 °C by the isolated hydrogen atoms in infrared spectroscopy (Ref. 12). Furthermore, one recognizes that even preformed aromatic rings are splitted off in case of steric hindering. The practical application of parapolyphenylene as precursor, however, is limited because of its poor processability. This polymer is non-melting and also insoluble in all solvents.

A modification of the polyphenylene structure by isomerics (Ref.13) results in synthetic pitches with lowered carbon yield. Also the precondition of unmeltability of the polymers is lost, and an additional stabilization treatment is needed before carbonization.

Fig. 9. Suitable polymers for the formation of isotropic polymeric carbon.

Some of the most suitable thermosetting polymers which have been studied as precursor for polymeric carbon are summarized in Fig.9. Besides phenolics with various hardeners also polymides, f.i. "kaptone" foils can be used for fabrication of polymeric carbon sheets (Ref. 14). The thermosetting resin HA 43 yields 85% carbon, because it has a polyphenylene structure. It is cured by its endstanding acethylene groups (Ref. 15).

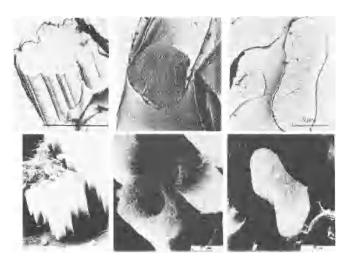
Fig. 10. Stabilization of thermoplastic polymer fibres

Instead of using the cross linked thermosetting polymers as precursor material one can also start from thermoplastic chain-polymers in order to achieve a better forming into the desired shape, but has to transform the polymer afterwords into a thermostable ladder polymer. This procedure is applied in modern carbon fibre technology by the so-called stabilization treatment. Best known is the polyacrylonitrile precursor (PAN) with a continous carbon/carbon chain as backbone. PAN is stabilized by oxidation treatment in the 200 °C - area to heterocyclic naphthyridine ring systems which are dehydrogenated by oxygene into an aromatic ladder structure (Fig. 10). The planar polyaromatic polymers are formed by condensation reactions and splitting off the heteroatom nitrogen in the temperature range at about 600 °C. Other chain polymers like 1,2-poly-butadien or poly-isoprene should react in the same manner, but are less economic than PAN.

The stabilization treatment is the slowest and thus rate controlling step in the carbon fibre production because the needed oxygen must diffuse through the already stabilized outer layer. Therefore, much work has been performed on the catalysis of cyclization (Ref.16) and on the effect of copolymers during the stabilization treatment (Ref. 17).

Recently, it was shown that COURTELLE fibres containing 6 % polymethylacrv-late as co-polymer show a much higher stabilization rate as ORLON which has the same copolymer (Ref. 18). The high stabilization activity is devoted to an additional small content of itaconic acid as dying additive in COURTELLE. If however a prestabilization of ORLON is performed by heat treatment in vacuum, the same final oxidation rate as in COURTELLE can be observed, showing that mainly the outer oxidized shell reduces the stabilization rate.

At the moment it seems that nearly most of commercially available textile polyacrylo nitrile fibres can be used as precursor for the carbon fibre fabrication (Ref. 19). Only the optimum stabilization conditions have to be modified. As textile grade PAN fibres are unexpensive the use of petrol pitch as precursor for carbon fibres due to economic reasons becomes a less attractive alternative. In any case, the morphology of the precursor fibre is fully preserved. This can be seen in the micrographs (Fig. 11) for the case of rayon (left hand side). wet spun PAN (middle part) and for PAN produced in a dry spinning process (right part)



TEM OF

PRECURSOR FIBRES

SEM OF

CARBON FIBRES

Fig. 11. Morphology of precursor and carbon fibres (left rayon, middle part wet spun PAN) right part dry spun PAN)

Fig. 12 shows the structural model of anisotropic polymeric carbon from PAN (Ref.20). The polyaromatic carbon sheets in form of ribbons and plates are well oriented in the direction parallel with the fibre axes, but do not exhibit a preferred orientation perpendicular to it, so leading to a good compressive strength and sufficient resistivity against shear.

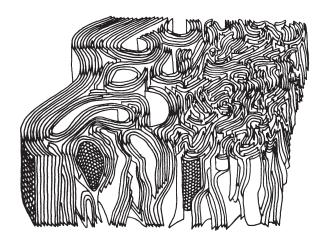


Fig. 12. Structural model of C-fibres prepared from PAN(Ref.20)

The formation of carbon fibres from pitch follows a different route from chemical as well as morphological viewpoint. Pitches form during heat treatment at 400 $^{\circ}$ C liquid crystals, a spherical mesophase (Fig. 13), (Ref. 21), which consists of planar polyaromatics as indicated in Fig. 14 (Ref. 22).



<u>50µm</u>

Fig. 13. Anisotropic mesophase, formed in isotropic petrol pitch by heat treatment at 400 $^{\circ}\mathrm{C}$.

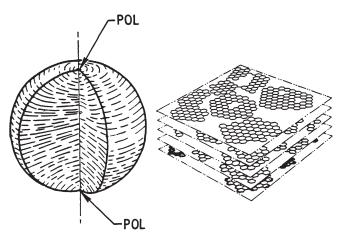


Fig. 14. Structural model of the liquid crystal-like mesophase which consists of planar polyaromatic molecules (Ref. 22).

The mesophase can coalescent, causing an increase of the viscosity, and finally formation of the anisotropic bulk-mesophase which is known to be the starting material for highly anisotropic and well graphitizing coke. If used for the fabrication of polymeric carbon in the form of carbon fibres it has to be stabilized after spinning by similar oxidation treatments as PAN, in order to perform a cross linkage between the planar sheets and to inhibit graphitization. The carbon fibres derived from pitch have a different structure and thus also different properties (Fig. 15).



Fig. 15. SEM of the fracture surface of a pitch carbon fibre

More important are the final mechanical properties of the various carbon fibre types. The PAN based carbon fibres are higher in strength, but the strength decreases by graphitization from $4000 \rightarrow 2500 \text{ MN/m}^2$, whereas the YOUNG's modulus increases from $250 \rightarrow 400 \text{ GN/m}^2$. The pitch fibres have a lower strength of 200 MN/m^2 only which is not affected by graphitization treatment. However, the YOUNG's modulus can increase more by graphitization than in PAN based carbon fibres, namely from $350 - 500 \text{ GN/m}^2$. Tensile strength and YOUNG's modulus of the various reinforcing fibres are

Tensile strength and YOUNG's modulus of the various reinforcing fibres are compiled in Fig.16. All C-fibre types exceed the organic high modulus fibres as far as the YOUNG's modulus is concerned, but only the type II which is finally heat treated below 1500 °C, reaches the high strength of glass and B-fibres. C-fibres type I are graphitized above 2500 °C, those of type III heat treated below 1200 °C only.

The graphitized C-pitch fibres are the best in YOUNG's modulus but lowest in strength.

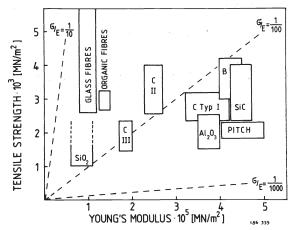


Fig. 16. Tensile strength and YOUNG's modulus of the various fibres used for reinforcement.

Fig. 17 shows average prices of the various fibres and also the cost of a volume unit strength and YOUNG's modulus for comparison (Ref. 23). As can be seen, the organic high modulus— and the glass fibres are the most unexpensive ones as far as the strength is concerned. Although also the cost for a YOUNG's modulus volume unit seems to be unexpensive, these fibres are not suitable for the fabrication of high modulus composites.

The strength unit of PAN-C fibres type II are quite unexpensive, whereas the even less expensive pitch carbon fibres result in higher costs for the volume unit of strength. This type seems to be advantageous in high modulus structures only.

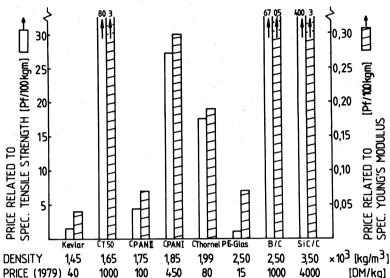


Fig. 17. Prices of the various fibre types and cost of a volume unit strength and modulus (Ref. 23).

APPLICATION OF POLYMERIC CARBON AS MONOLYTHIC MATERIAL

The isotropic form of polymeric carbon, the glassy carbon, is a corrosion resistant impervious material with a density of 1,5 kg/m 3 . The sample sizes which can be made are limited to some millimeters in thickness due to the irreversible shrinkage during thermal degradation.

This material is used in laboratory equipments, but mainly in human medicine because of its outstanding biocompatibility, f.i. as electrodes, percutanous leads, orologic protheses, dental and joint implants. The most striking success of monolythic carbon is achieved with its application as heart valve components, where glassy carbon is introduced instead of pyrocarbon for economic reasons (Ref. 24).

The anisotropic form of polymeric carbon, the fibres, can be applied for replacement of ligands and tendons (Fig. 18, Ref.25, 26).

Following some years of successful experiments with animals, this implantation technique begins to be transferred to human surgery. There is the possibility even to replace the abdominal wall by carbon fibres in form of waven textiles (Ref. 27).





Fig. 18. New ligand tissue is grown surrounding the implanted carbon fibres (Ref. 24).

APPLICATION OF CARBON FIBRES FOR REINFORCEMENET OF POLYMERS

The most important application of anisotropic polymeric carbon is that reinforcing fibres in composites, but only some of the most important facts can be referred in this introductory paper.

A fundamental parameter which controls the final properties of the composite is the geometric arrangement of the reinforcing fibres within the polymer matrix. Fig. 19 (Ref.28) shows the effect of the fibre direction on the strength of the composite for the case of a two-directional reinforcement. The precalculation is based on a modified rule of mixture, taking into consideration volume fractions and mechanical properties of the components as well as the direction of fibre arrangement.

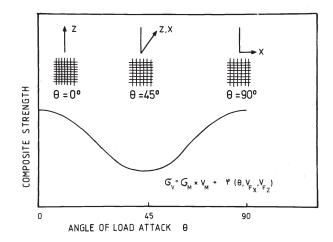


Fig. 19. Composite strength for two-directional reinforcement (Ref. 28).

The degree of adhesion between fibre and matrix is another important parameter which controls the mechanical behaviour of the composite. We have been able to show that this adhesion is mainly caused by chemical bonds between the fibre surface and the matrix (Ref. 29). The reactive surface groups on the carbon fibres play a dominant role for this chemical bonds. If one uses a surface treated fibre which has been oxidized, various types of surface oxides are present.

$$\begin{array}{c|c} & & & \\ &$$

Fig. 20. Chemical reactions between the acid surface groups on oxidized C-fibres and the hydroxil groups of the phenolic matrix (Ref. 29).

Fig. 20 for instance shows the chemical reactions between carboxilic groups on the fibre surface and the hydroxil groups of the phenolic matrix (Ref.29). Similar reactions are known between acid and basic surface groups on C-fibres and epoxy matrix (Ref. 30, 31).

Finally, the fracture behaviour of the matrix material is of great impor-tance. Fig. 21 indicates the stress/strain behaviour for ductile and for brittle matrices. Full utilization of the fibre strength can only be obtained if the strain to failure of the matrix exceeds that of the fibre (left part of Fig. 21). Otherwise, the ultimate strength of the composite will be controlled by the strain to failure of the matrix as indicated in the right part of Fig. 21. In such cases, fibres with highest YOUNG's modulus are most advantageous (Ref. 32). This situation will be described with carbon/carbon composites.

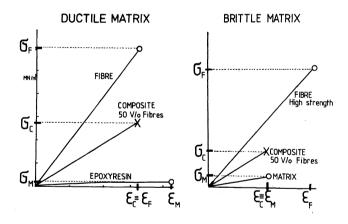


Fig. 21. Stress-strain behaviour of fibre reinforced composites with ductile and brittle matrix respectively (Ref. 32).

The most important applications are C-fibre reinforced polymers. These composites which correspond to the combination brittle fibre with ductile matrix, were introduced into the modern technology by the need of supersonic aircrafts and space vehicles for extremely stiff but very light new materials for structures. At the moment an additional tendency to replace conventional metal parts by such composites in motive structures like cars can be observed in order to save weight and energy. The car industry in the western countries try to use fibre reinforced composites in the body structure, frame doors, front end assembly, drive shaft, bumper, wheels, springs, brakes and other parts. Even inner parts of the engine can be replaced by composites (33). More than 50 kg carbon fibres could be used in each car. It was calculated that a

considerable amount of energy can be saved not only during the fabrication, but even more during the use of the car by less fuel consumption, if metals are replaced by carbon/epoxy composites. Figure 22 (Ref.34) shows the results of such precalculations for a 5 year life time of the components.

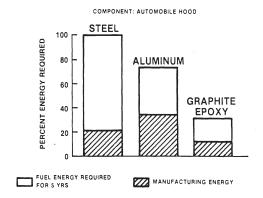


Fig. 22. Component life time energy savings with composites, replacing metals in future cars (Ref. 34).

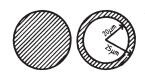
There remains no doubt, that carbon fibre reinforced polymers (CFRP) with their extremely high stiffness (up to $30~\rm{GN/m^2}$) and low density (1,5 kg/m³) will replace metals in the most future applications where weight saving is the main goal. The problem is only that manufacturing of f.i.cars from metals is so highly rationalized in industrialized countries, whereas the present production methods of composites still causes high labour costs. The composite technology offers therefore a great chance for development countries with high labour capacity.



Fig. 23. Surgery tool made of CFRP offers in situ X-ray control during operation.

Also in human surgery, C-fibre reinforced composites will play an important role in future. Bone plates in osteosynthesis are one example (Ref. 35) because of the possibility to tailor the mechanical properties and thus to match the special need controlled by the elastic behaviour of bones. In case of CFRP as material for tools in surgery as shown in Fig. 23 (Ref. 36) one utilizes the additional advantage that neither carbon nor polymers absorbe X-rays and improved X-ray images can be taken in situ during operation.

Recently, other forms of reinforcing elements made of polymeric carbon have been developed, namely the hollow fibres (Ref. 37). Some precalculations of stiffness characteristics of composites with hollow fibres are compiled in Fig. 24 (Ref.38), and one recognizes that the lowered bulk density especially increases the bonding stiffness and the plate buckling strength, properties which offer not only further weight savings, but also improvement and simplification in design and manufacture of airplane structures.



STIFFNESS CHARACTERISTICS:

1. Extensional stiffness	<u>E</u> [mm]	8.0×10 ⁹	6.3×10 ⁹
2. Bending stiffness (Column buckling strength	$\int_{0}^{2\sqrt{E}} \frac{\sqrt{E}}{\rho g} \left[\frac{mm^2}{N^2} \right]$	2.3×10 ⁴	2.6×10 ⁴
3. Plate buckling strength	$\frac{\sqrt[3]{E}}{\rho g} \begin{bmatrix} \frac{7}{mm^3} \\ \frac{2}{N^3} \end{bmatrix}$	3.3×10 ³	4.0×10^3

Fig. 24. Stiffness characteristics of composites which show the advantage offered by hollow fibres as reinforcing element (Ref. 38).

We succeeded in preparation of hollow carbon fibres (Fig.25) from wet spun hollow PAN-precursor fibres in a continously working laboratory equipment as miniplant. The carbon material of the hollow fibres has similar mechanical properties as that of type III C-fibres.

But this development is only in a very early experimental stage.



Fig. 25. SEM of a C-hollow fibre from PAN precursor (Ref.37)

Finally, another form of composites should be referred, namely the combination of isotropic and anisotropic polymeric carbon in one combined all-carbon material, the carbon/carbon-composites. This material was developed in the past mainly for military reasons as conetip-and nozzle material in big missiles (Ref. 40).

However, carbon/carbon - composites offer such interesting combinations of properties that they will become of much more interest for many other applications. Results of basic studies on unidirectionally carbon fibre reinforced all-carbon composites (CFRC) can be found in the newer literature (Ref. 41 - 46). If isotropic matrix-carbon is used, up to 100 % utilization of the introduced fibre strength can be achieved by optimization of the complex liquid impregnation process.

If the carbon matrix itself is prepared with preferred orientation in direction parallel with the fibres (Fig. 26) it will contribute to the bulk modulus in the same amount as the fibre, and composites with stiffness of $400~\mathrm{GN/m^2}$ in fibre direction have been achieved (Ref. 47). Fig. 26 exhibits the sheetlike structure of the carbon matrix.

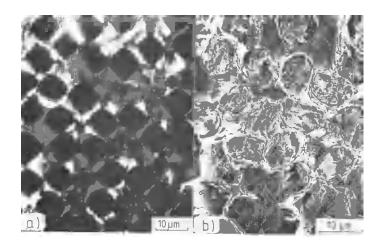


Fig. 26. Micrographs of unidirectionally reinforced carbon/carbon composites with anisotropic carbon matrix, a) polarized light after polishing, b) SEM after etching by oxidation in air.

The achievable bulk data in strength and stiffness of carbon/carbon composites with various arrangement of the reinforcing carbon fibres are compiled in Fig. 27.

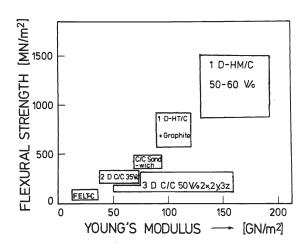


Fig. 27. Mechanical bulk properties of carbon/carbon composites [1,2,3-D] indicates directions of reinforcement, HM= type I, HT= type II, C/C sandwich was prepared by C-fibre textiles (Ref. 46%).

The variability in geometrical arrangement offers the possibility for tailoring the properties of the composite. This f.i. is most important for medical application as implantation material. One of the most serious problems in implantations surgery is that of the hip joints replacement.

The present solution for replacement of damaged hip joints with the steel stem and a polyethylene soket is not finally satisfying. As known the average life time of such an implantation is 5 years only. An improvement is expected by the introduction of carbon materials, as shown in Fig. 28. These developments are strongly supported by the German government (Ref. 48). The friction and abrasion problem between ball and socket seems to be soluble by the material combination of fine grained graphite with SiC impregnated graphite (Ref. 49).

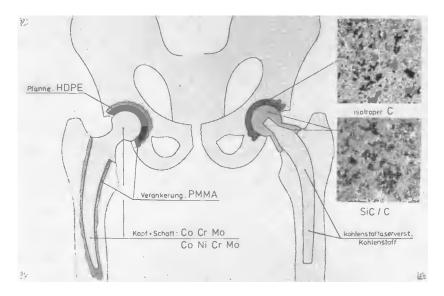


Fig. 28. Material solutions for hip joint implantages left side : present solution, right side: carbon solution (Ref.49).

The strength problem of the stem is hoped to be controlled by tailored carbon/carbon composites. One possibility of solution is indicated in Fig.29 (Ref. 24). It can also be recognized that intramedullary screw-thread is foreseen for avoiding the unsatisfactory use of bone cement (Ref. 50).

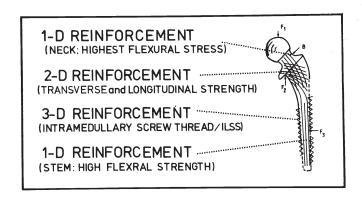


Fig. 29. Possibilities offered by carbon/carbon composites to imitate the femur structure for application as stem material in hip joint implantations (Ref. 50).

It is hoped having shown that an enormeous field of application can be expected for polymeric carbon, in isotropic as well as in anisotropic form, and in various combinations of both. Perhaps the reader will agree that the way from polymers to polymeric carbon could be the key for a large variety of future new materials. Polymeric carbon will not replace or substitute the polymers, but it will improve these materials in a growing number of applications.

REFERENCES

- 1. H. Kindler, A. Nikles, CIT, 51 H 11 p.1 (1979).
- 2. G.M. Jenkins, K. Kawamura, Polymeric Carbon, Cambridge University Press, (1976).
- 3. W. Braun, <u>Thesis</u>, University of Karlsruhe, W.-Germany (1974). 4. F. Rousseaux, D. Tschoubar, <u>Carbon</u>, <u>15</u>, pp. 63-68 (1977). 5. E. Fitzer, J. Kalka, <u>Carbon</u>, <u>10</u>, pp. 173-183 (1972).
- E. Fitzer, W. Hüttner, Abstr. Conf. on Strength and Structure in Carbons and Graphite, paper 21, University of Bath (1979).
- 7. D. Overhoff, Thesis, University of Karlsruhe, W.-Germany (1966).

- J.A. Mock, Mat. Design. Eng., 49, S. 149, (1959).
 E. Fitzer, W. Schäfer, Carbon, 8, pp 353 364 (1970).
 E. Fitzer, K. Müller, W. Schäfer, Chemistry and Physics of Carbon, ed. Walker, 7, pp 237-383 (1971).
 F. Rousseaux, Thesis, University of Orléans, France (1975).
 E. Fitzer, J. Kalka, High Temperatures High Pressures, 3, p.53 (1971).
 E. Fitzer, F. Grieser, Proc. 5th London Int. Carbon and Graphite Conf. Vol. I, pp 266-279 (1978).
 A. Bürger, E. Fitzer, M. Heym, B. Terwiesch, Carbon, 13, pp 149-157 (1975).
 E. Fitzer, K.-H. Geigl, W. Hüttner, Proc. 5th London Int. Carbon and Gra-
- 15. E. Fitzer, K.-H. Geigl, W. Hüttner, Proc. 5th London Int. Carbon and Graphite Conf., Vol. I, pp 493-506 (1978).

 16. E. Fitzer, M. Heym, Chemistry and Industry, 16, pp 663-676 (1976).

 17. J. Simitzis, Thesis, University of Karlsruhe, W.-Germany (1975).

- 18. W. Johnson, W. Watt, Abstr. Strength and Structure in Carbons and Graphite University of Bath (1979), compare: Nature, Vol. 257, pp 210-212 (1975).

 19. E. Fitzer, main lecture at Hauptversammlung GdCh, Berlin (1979), in print.
- Angew. Chemie, no 5, vol.92 (1980) and Angew. Chemie, International Ed. Engl. no 5, vol. 19 (1980).
- 20. S.C. Bennett, D.J. Johnson, Proc. 5th London Int. Carbon and Graphite Conference, Vol. I, pp 377-386 (1978).
 21. J.D. Brooks, G.H. Taylor, Chemistry and Physics of Carbon, ed. Walker, Vol. 4, p 243 (1968).

- Vol.4, p 243 (1966).

 22. J.L. White, Aerospace Report, Nr. TR 0074 (4250-40), 1 (1974).

 23. R. Diefendorf, E. Fitzer, M. Heym, CIT, 48, p 765-773 (1976).

 24. E. Fitzer, W. Hüttner, D. Wolter, 3rd Int.Symp. on Newer Fibres and Composites, Bombay, Feb. 1978, compare Man Made Textiles, 8, p 389 (1978).

 25. D.H.R. Jenkins et al., J. of Bone and Joint Surgery, 59/B, p 53 (1977).

 26. E. Fitzer, W. Hüttner, D. Wolter, Abstr.13 th Bienn.Conf. on Carbon,
- Irvine, USA, pp 180-181 (1977).
- 27. R. Neugebauer, C. Claes, G. Helbing, D. Wolter, 11th Int.Biomat.Symp.,
- 5th Ann. Meet.Soc. Biomat. 3, p 135 (1979).

 28. C.D. Ellis, B. Harris, J. of Composite Material, 7, p 76 (1972)

 29. E. Fitzer, K.-H. Geigl, L.M. Manocha, Proc. 5th London Int. Conf. on
- Carbon and Graphite, Vol. 1, p 405, (1978)

 30. K. Horie, H. Murai, I. Mita, Fibre Science and Technology, 9, p 253 (1976).

 31. E. Fitzer, K.-H. Geigl, W. Hüttner, R. Weiß, Abstr. 14th Bienn.Conference on Carbon, Pennsylvania State Univ. pp 220-221 (1979), compare Carbon, in print.
- 32. E. Fitzer, Proc. of Int.Symp. of Factors in Densific. and Sint. of Oxides and Non-oxide Ceramics, Hakone, Japan, pp 618-673 (1978).

 33. New Release, Ford Motor Company (October 1977)
- 34. G.C. Deutsch, 23th SAMPE Symp. and Exhibition, Anaheim, Ca., USA, p 34 (1978).
- E. Fitzer, W. Hüttner, L. Kinzl, L. Claes, Preprint 14th Biennial Conf. on Carbon, Pennstate University, p 58 (1979).

- 36. P. Wolter, P. Mattes, G. Helbing, <u>Der Chirurg</u>, 49, p 253 (1978).

 37. K.-H. Geigl, <u>Thesis</u>, University of Karlsruhe, W.-Germany (1979).

 38. W. Bergmann, <u>Abstr. Newer Frontiers in Polym.Science and Polymer Appl.</u> Madras, India (1980)
- 39. C.G. Goetzel, <u>High Temperatures-High Pressures</u>, in print 40. L.E. McAllister, A. Taverna, <u>Prepr. 1oth Bienn.Conf. on Carbon</u>, <u>Bethlehem</u>, L.E. McAllister, A. Taverna, <u>Frepr. 10th Stem.</u>.
 USA, paper FC 40, (1971).
 M. Heym, <u>Thesis</u>, University of Karlsruhe, (1976)
 E. Fitzer, K.-H. Geigl, W. Hüttner, <u>Proc. 5th London Intern. Conf. on Carbon and Graphite</u>, Vol. 1, p 493 (1978).
 C.R. Thomas, E.J. Walker, <u>High Temperatures-High Pressures</u>, 10, p 79(1978).
 E. Fitzer, K.-H. Geigl, W. Hüttner, <u>Prepr. 14th Biennial Conf. on Carbon</u>, <u>Pennstate Univ.</u>, p 236, (1979), compare <u>Carbon</u>, in print.
 F. Fitzer, W. Hüttner, L.M. Manocha, <u>Prepr. 14th Biennial Conf. on Carbon</u>,

- 45. E. Fitzer, W. Hüttner, L.M. Manocha, Prepr. 14th Biennial Conf. on Carbon, Pennstate Univ., p 240, (1979), compare Carbon, in print.
- 46. E. Fitzer, W. Hüttner, Proc. 93. Wehrtechnisches Symposium, Mannheim, (Mai 1978).
- 47. W. Hüttner, Thesis, University of Karlsruhe (1980).
- 48. Research Project OI VG 118 A ZK/NT/MT 290, supported by German Government BMFT
- 49. H. Brückmann, Thesis, University of Karlsruhe, W.-Germany (1979).
- 50. D. Wolter, private Communication, (1978).